## Hall Ticket Number:

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## III/IV B. Tech (Regular) DEGREE EXAMINATION

OCTOBER, 2016

Third Semester
Time: Three Hours
Answer Question No. 1 compulsorily.
Answer ONE question from each unit.

1. Answer all questions

## Chemical Engineering

Momentum Transfer

Maximum : 60 Marks
(1X12 = 12 Marks)
(4X12=48 Marks)
(1X12=12 Marks)

1. Explain the following:

| a | Define unit processes. Any chemical Change. |
| :---: | :---: |
| b | Write the equation to calculate separation time in gravity decanter $100 . \mathrm{mu} /\left(\right.$ rhoA-rhoB) where mu is in cPa and rho is in $\mathrm{kg} / \mathrm{m}^{3}$ and t is in Hr . |
| c | Define boundary layer <br> Part of fluid influenced by solid boundary interms of shear rate and strear stress gradients. |
| d | Define mass velocity Velocity multiplied by density. |
| e | What in the relation between velocity and radius in a horizontal pipe line. <br> For a Newtonian and incompressible fluid under steady state-isothermal flow velocity varies parabolically with radius in laminar flow. And in turbulent flow exponentially. |
| f | Define kinetic energy correction factor. $\frac{1}{S} \int\left(\frac{u}{\bar{V}}\right)^{3} d s$ |
| g | Write continuity equation for compressible fluid $d(\rho u S)=0$ |
| h | Define drag coefficient $\frac{F_{D} / A_{p}}{[1 / 2] \rho u^{2}}$ |
| i | Write Burke-Plummer equation $\frac{\Delta P}{L}=1.75 \frac{\rho \bar{V}_{0}^{2}}{\emptyset_{S} D_{P}} \frac{(1-\varepsilon)}{\varepsilon^{3}}$ |
| j | Pieces of tubing are connected by Soldering, compression fittings... |
| k | What are different forces acting in Rotameter |


|  | Drag, gravity and buoyancy. |
| :---: | :--- |
| 1 | Classify positive displacement pumps. <br> Rotary and reciprocating |

## UNIT - I

| 2.a | Explain Rayleigh's method of dimensional analysis. <br> Rayleigh's method is based on dimensional homogeneity. Therefore dependent <br> variable is expressed as product of independent variables raised to empirical <br> powers. The equations obatained by dimensional homogeneity are solved. <br> Regrouping is done to obtain reduced number of groups which are dimensionless. | 3 x 2 <br> 6 M |
| :--- | :--- | :--- |
| 2.b | With the help of neat sketch explain the principle of U-tube manometer <br> Manometer is an important device for measuring pressure difference. With <br> respect to above figure applying hydrostatic equilibrium at plane 2-3. | $3 \mathrm{x}=6 \mathrm{M}$ |
| $p_{a}+\frac{g}{g_{c}}\left[\left(Z_{m}+R_{m}\right) \rho_{B}-R_{m} \rho_{A}-Z_{m} \rho_{B}\right]=p_{b}$ |  |  |
| Simplification of this equation gives... |  |  |
| $p_{a}-p_{b}=\frac{g}{g} R_{m}\left(\rho_{A}-\rho_{B}\right)$ |  |  |

## Or

| 3.a | Explain boundary layer formation in straight tubes and horizontal plates In straight tubes boundary layer reaches to the center of tube in laminar flow for incompressible Newtonian fluid. A very thin boundary layer is formed in case of high turbulent flows. In between these flows boundary layer spreads from solid wall to anywhere near center of tube. | $3+3 \mathrm{M}$ |
| :---: | :---: | :---: |
| b | Show that in a gravity decanter the position of the liquid - liquid interface in the separator depends on the ratio of the densities of the two liquids and on the elevation of the overflow line. <br> Applying principle of hydrostatic equilibrium $Z_{B} \rho_{B}+Z_{A 1} \rho_{A}=Z_{A 2} \rho_{A}$ <br> Solving for ZA1 gives... $Z_{A 1}=Z_{A 2}-Z_{B} \frac{\rho_{B}}{\rho_{A}}=Z_{A 2}-\left(Z_{T}-Z_{A 1}\right) \frac{\rho_{B}}{\rho_{A}}$ <br> Interms of $\mathrm{Z}_{\mathrm{T}}$ $Z_{A 1}=\frac{Z_{A 2}-Z_{T}\left(\rho_{B} / \rho_{A}\right)}{1-\rho_{B} / \rho_{A}}$ | $\begin{aligned} & 3 \times 2=6 \\ & \mathrm{M} \end{aligned}$ |


| 4.a | a) Derive an expression for velocity distribution in a horizontal pipeline. <br> And for any element of general radius $r$ it follows that $\frac{d p}{d L}+\frac{2 \tau_{r z}}{r}=0$ <br> Therefore the shear stress profile is $\frac{\tau_{r z}}{r}=\frac{\tau_{w}}{r_{w}}$, which is a linear shear stress profile. <br> $\tau_{r z}=0$ at center of pipe and $\tau_{r z}=\tau_{w}$, the maximum value at the wall. <br> Integrating the above differential equation and solving for integrating constant the <br> velocity profile is $V_{z}=\frac{\tau_{w} r_{w}}{2 \mu}\left[1-\left(\frac{r}{r_{w}}\right)^{2}\right]$ <br> Therefore $V_{z}$ at $r=0$ is equal to $V_{z, m a x}=\frac{\tau_{w} r_{w}}{2 \mu}$ | $2+2+2$ <br> $=6 \mathrm{M}$ |
| :--- | :--- | :--- |
| 4.b | Derive an expression for shear stress distribution in cylindrical tube of incompressible flow <br> of fluids. <br> And for any element of general radius $r$ it follows that $\frac{d p}{d L}+\frac{2 \tau_{r z}}{r}=0$ | $2+2+2$ <br> $=6 \mathrm{M}$ |
|  | Therefore the shear stress profile is $\frac{\tau_{r z}}{r}=\frac{\tau_{w}}{r_{w}}$, which is a linear shear stress profile. <br> $\tau_{r z}=0$ at center of pipe and $\tau_{r z}=\tau_{w}$, the maximum value at the wall. |  |

Or

| 5.a | Derive the relation between average velocity and maximum velocity in a horizontal pipeline <br> Integrating the above differential equation and solving for integrating constant the velocity profile is $V_{z}=\frac{\tau_{w} r_{w}}{2 \mu}\left[1-\left(\frac{r}{r_{w}}\right)^{2}\right]$ <br> Therefore $V_{z}$ at $r=0$ is equal to $V_{z, \max }=\frac{\tau_{w} r_{w}}{2 \mu}$ <br> Expression for cross sectional average of velocity $\bar{V}_{Z}$ can be derived as follows $\bar{V}_{z}=\frac{\int_{0}^{S} V_{z} d S}{\sum_{0}^{S} d s}=\frac{1}{\pi r_{w}^{2}} \sum_{0}^{r_{w}} \frac{\tau_{w} r_{w}}{2 \mu}\left[1-\left(\frac{r}{r_{w}}\right)^{2}\right] 2 \pi r d r=\frac{\tau_{w} r_{w}}{4 \mu}$ <br> The it follows that $\frac{\bar{V}_{Z}}{V_{z, \text { max }}}=0.5$ | $\begin{aligned} & 2+2+2=6 \\ & M \end{aligned}$ |
| :---: | :---: | :---: |
|  | In a natural gas pipe line at station 1 , the pipe diameter is 0.61 m and the velocity is $15 \mathrm{~m} / \mathrm{s}$ and the density is $39 \mathrm{~kg} / \mathrm{m}^{3}$. At station 2, the pipe diameter is 0.914 m and the density is $24 \mathrm{~kg} / \mathrm{m}^{3}$. Calculate the velocity at station 2 and the mass flow rate. $(0.61 / 0.914)^{2} \times 39 / 24 \times 15=\mathrm{U} ; \text { MFR=39x15x3.14x0.61x0.61/4 = } 170.88 \mathrm{~kg} / \mathrm{s}$ | 3+3=6M |

## UNIT - III

| 6.a | Derive steady flow total energy balance equation for the flow of compressible fluids Assumptions <br> 1. Steady state flow. <br> 2. 1-D flow. <br> 3. Gravity effect negligible. <br> 4. $\mathrm{C}_{\mathrm{p}}$ are constant and $\mathrm{Cp} / \mathrm{Cv}$ is constant. | $\begin{aligned} & 6 \times 1= \\ & 6 \mathrm{M} \end{aligned}$ |
| :---: | :---: | :---: |


|  | 5. Internal fluid friction is negligible. <br> 6. No shaft work. Total Energy Balance Equation $d H+d\left(\frac{u^{2}}{2}\right)+g d Z=d Q$ |  |
| :---: | :---: | :---: |
| 6.b | Derive Ergun equation for flow through beds of solids. $\begin{gathered} \bar{V}=\frac{\bar{V}_{O}}{\varepsilon} \\ n\left[\frac{\pi}{4} D_{e}^{2} L\right]=S L \epsilon \\ n\left[\pi D_{e} L\right]=\frac{6}{\emptyset_{S} D_{P}} S L(1-\varepsilon) \\ D_{e}=\frac{2}{3} \emptyset_{S} D_{P} \frac{\varepsilon}{(1-\epsilon)} \\ \frac{\Delta P}{L}=\frac{72 \lambda_{1} \mu \bar{V}_{0}}{\emptyset^{2} D_{P}^{2}} \frac{(1-\varepsilon)^{2}}{\varepsilon^{3}} \\ \frac{\Delta P}{L}=3 f \lambda_{2} \frac{1}{\emptyset_{S} D_{P}}\left(\rho \bar{V}_{0}^{2}\right) \frac{(1-\varepsilon)}{\varepsilon^{3}} \\ \frac{\Delta P}{L}=\frac{150 \mu \bar{V}_{0}}{\emptyset^{2} D_{P}^{2}} \frac{(1-\varepsilon)^{2}}{\varepsilon^{3}}+1.75 \frac{\rho \bar{V}_{0}^{2}}{\emptyset_{S} D_{P}} \frac{(1-\varepsilon)}{\varepsilon^{3}} \end{gathered}$ | $\begin{aligned} & 6 \times 1= \\ & 6 \mathrm{M} \end{aligned}$ |

Or

|  | Explain different processes of compressible flow. <br> 1. Isentropic flow: Reversible adiabatic flow. Therefore stagnation temperature is constant along flow. The conduit in which isentropic flow takes place is known as a 'nozzle'. <br> 2. Irreversible adiabatic flow: Also known as adiabatic frictional flow | $\begin{aligned} & 2+2+ \\ & 2=6 \\ & \mathrm{M} \end{aligned}$ |
| :---: | :---: | :---: |

3. Isothermal frictional flow: Temperature throughout the flow is constant.
Constant cross sectional area conduit is considered. Flow is neither
isentropic nor adiabatic. Stagnation temperature is varying
Reservoir

## UNIT - IV

| 8.a | a) Explain working principle of centrifugal pump. <br> The discharge connection is mounted concentrically with the eye of the high <br> speed rotary device called impeller. Radial vanes are cast integrally in the <br> impeller. Fluid that enters the eye of the impeller due to the vacuum created by <br> the impeller by emptying the fluid in the pump by centrifugal action fills the space <br> available between the vanes and leaves the periphery of the impeller at a higher <br> velocity compared to that at the entrance of the impeller. For a well designed <br> pump the space between the vanes is completely filled by the liquid and there is <br> no cavitation. The fluid leaving the impeller periphery is collected in the spiral <br> casing called volute. Fluid leaves though the discharge connection at higher <br> pressure. The kinetic energy of the fluid is converted to pressure energy in the <br> volute. The torque applied to shaft of the impeller imparts kinetic energy to the | M |
| :--- | :--- | :--- |


|  | fluid through the impeller. |  |
| :--- | :--- | :--- |
| 8.b | A horizontal venturi meter having a throat diameter of 20 mm is set in a 75-mm-ID <br> pipeline. Water at $15{ }^{0} \mathrm{C}$ is flowing through the line. A manometer containing mercury <br> under water measures the pressure difference over the instrument. When the manometer <br> reading is 500 mm. what is the flow rate in $\mathrm{m}^{3} / \mathrm{h}$ ? | $2+1+1=$ <br> 4 M |
| $Q=C_{d} \frac{a_{1} a_{2}}{\sqrt{a_{1}^{2}-a_{2}^{2}}} \times \sqrt{2 g h} \quad \mathrm{Cd}=0.98$ assumed; $\mathrm{a}_{1}=3.14 / 4 \times 0.02 \times 0.02=$ |  |  |
| $\mathrm{a}_{2}=3.14 / 4 \times 0.075 \times 0.075=$ <br> $\mathrm{g}=9.81 \mathrm{~m} / \mathrm{sq.s} ; \mathrm{h}=0.5(13.6-1) \mathrm{m} ;$ |  |  |

## Or

| 9.a | Explain working principle of orifice meter <br> The principle of the orifice meter is identical with that of the venturi. The reduction of the cross section of the flowing stream in passing through the orifice increases the velocity head at the expense of the pressure head, and the reduction in pressure between the taps is measured by the manometer. Bernoulli's equation provides a basis for correlating the increase in velocity head with the decrease in pressure head. | $\begin{aligned} & \mathrm{Fig}=2 \\ & \mathrm{M} \\ & 4 \mathrm{M} \end{aligned}$ |
| :---: | :---: | :---: |
|  | Explain working principle of pitot tube. <br> The pitot tube is a device to measure the local velocity along a streamline. The principle of the device is shown in Fig. 8.27. The opening of the impact tube $a$ is perpendicular to the flow direction. The opening of the static tube $b$ is parallel to the direction of flow. The two tubes are connected to the legs of a manometer or equivalent device for measuring small pressure differences. The static tube measures the static pressure Po since there is no velocity component perpendicular to its opening. | $\begin{aligned} & \text { 4x1=4 } \\ & \text { M; } \\ & \text { Fig=2 } \\ & \text { M } \end{aligned}$ |



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